

Advisory | Circular

Subject:

THE CONTINUED AIRWORTHINESSS
OF OLDER AIRPLANES

Date: 6/1/8/883

Initiated by: AWS-3340

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- 1. PURPOSE. This advisory circular (AC) provides information and recommendations to manufacturers and airplane owners operators for the development and use of programs to assure the continued airworthiness of older airplanes not covered under AC 91-56, Supplemental Structural Inspection Program for Large Transport Category Airplanes.
- 2. Airplanes are designed and built to provide many years of service. For the airplane to remain airworthy and safe to operate for a long inservice life, it should be operated in accordance with the recommendations of the manufacturer and cared for with sound inspection and maintenance practices.
 - a. Service experience has revealed that an aging **airplane** needs **more** care and special attention during the maintenance processes and, at **times**, requires more frequent inspection of structural components for damage due to **environmental** deterioration, accidental **damage**, and fatigue. **Typical**areas requiring **more** frequent inspect ion, and key to continuing the useful life of an airplane are structural points such as:
 - (1) Wing spar capstrips, with particular attention to the spar lower capstrips.
 - (2) Wing attach points, both on the wing and on the fuselage.
 - ((3)) Wing fuselage carry-through structure.
 - (4) Wing upper and lower covers between spars.
 - (5) Pressurized structures, especially around doors, windows, windshields and other cutouts on pressurized airplanes.
 - (6) Horizontal tail spars and spar capstrips.
 - (7) Horizontal tail spar attachments and fuselage attachments.
 - (8) Horizontal tail fuselage carry-through structure.
 - (9) Wertifical tail spars and spar capstrips.
 - (10) Wattingal tail spar attachments and fuselage attachments.

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- (11) Lifting surfaces.
- (12) Control surfaces and their respective hinge points.
- (B) Control surface balance weight attachments.
- (14) Control system attach points at the various cokurol surfaces,
- (15) Tab hinges.
- (16) Tab actuators.
- (17) -Main landing gear attach points where landing gear loads feed into primary wing and/or fuselage structure.
- (18) Structure and flooring especially in the vicinity of lavatory and galley areas and areas of overboard drains and vents.
- (19) Structural and flooring area **attachments** in cabin **and** cockpit and at seat and **equipment** attachments.
- b. In addition to the key items outlined above which are important to safety \dot{of} flight, there are other areas that should receive special attention. Typical **areas are:**
 - (1) Engine mounts.
 - (2) Propeller blades.
 - (3) Door latching machanisms on pressurized airplanes.
 - (4) Windshields and windows on pressurized airplanes.
 - (5) Landing gear shirmy dampers.
 - (6) Main and nose gear supporting structure.
- (7) Cowling, fairing, and fillet attrachments and supporting structure ${\bf I}$
- (8) Materials used in aircraft compartment interiors (including finishes or decorative surfaces applied to the materials) to assure they have not deteriorated due to aging, cleaning or other causes that would increase their flamability over what it tiss when they were new.
- 3. CONDINUEDDAIRWORTHNINESS. To provide for a safe operation, a program called kcontinued airworthiness" should be developed by the manufacturer and used by convers/Operators where virtually every component empisising an airplane is involved in some form of preservation, inspection, maintenance, preventive maintenance, overhaul, repair, and/or replacement activity.

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a. Safety of **eperation** through continued airworthiness demands increasing vigilance as an airplane ages. Airplane structural materials do have finite lives, and the extent of these is affected by **age**, operational **environment**, and operational **experience** which the material endures in day-to-day usage of the airplane.

- bl Maintenance information needs to be continually updated. Open communication should exist with the cwner/operator advising the manufacturer as soon as a new situation arises, and the manufacturer responding with solid help. Such communication and cooperation will facilitate the maintenance of an entire fleet of airplanes in a continuously airworthy condition.
- 4. MANUFACTURERS PARTICIPATION. The framework for continued airworthiness of airplanes is dependent upon the care and thought designed into the airplane at its inception and reflected in the type design. Following type certification, the establishment of, and adherence to, thorough maintenance procedures is required. Thus the manufacturer should prepare and distribute recommendations on the need for increased inspection vigilance for those older (geriatric) airplanes which may not have a well-defined program for continued airworthiness, and which include areas of special consideration as are listed in paragraph 2(a) and (b) of this AC. This information should be available in clear, concise language to those who need it. Sometimes, especially for complex airplanes, it may be recessary to provide specific training in the proper maintenance of the airplane and its systems. These activities should be developed by the manufacturer based upon its familiarity with the designand function of the airplanes.
- 5. IMPLEMENTATION BY CONNERS/OPERATORS. Vigilance is the price of safety and, even though the manufacturers have done their best to anticipate all the maintenance required to provide for continued airworthiness, aging airplanes demand ever increasing care. Because of this the final care rests primarily with the owner/operator who should:
- a. Follow the manufacturer's **recommended** maintenance and inspection **produces.**
- b Recognize that **corrosion** or damage to structure can drastically shorten fatigue life and **be** on the lookout for these conditions.
- c. Be alert to the possibility that the airplane is not being used in a manner significantly different from the originally intended mission profile.

 Low altitude operation, such as pipeline patrol, sightseeing, or training operations, will subject the airplanes to more fatigue damage than high altitude cruise.
- d **Comply** with applicable Airworthiness Directives, and review possible modifications available **from** the **manufacturer.** Structural **improvements** are usually directed to lowering working stresses, and the sooner they are made, the better they are to slow the onset of **cumulattive** damage to the original **structure**.

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e. Become interested in the fatigue program on prospective new airplanes, including a clear understanding of the Elated inspection program recommended by the manufacturer, and make extraim all significant areas of the airplane's structure are treated in the airplane's inspection program.

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Adting Director of Airworthiness

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